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# **FAA APPROVED**

# SUPPLEMENTAL AIRPLANE FLIGHT MANUAL **FOR CESSNA 185 SERIES FLOATPLANES**

Equipped With AEROCET 3500 or 3500L Seaplane Floats

Registration No	
Serial No.	

The information contained in this document is FAA approved material which must be applied together with the basic FAA approved airplane placards and markings and/or FAA approved Airplane Flight Manual. This supplemental manual must be carried in the airplane when it is modified by the installation of the Aerocet Model 3500 or 3500L seaplane floats in accordance with Supplemental Type Certificate (STC) No. SA5908NM. The information contained in this document supersedes the basic airplane markings and placards and/or Flight Manual covered in the items contained herein. For Limitations, Procedures, and Performance information not contained in this supplement, consult the basic airplane markings and placards, and/or Flight Manual.

FAA Approved:

Manager, Special Certification Branch Seattle Aircraft Certification Office



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# LOG OF REVISIONS PAGE

REV.	PAGES AFFECT.	DESCRIPTION	FAA APPROVED
A	1,3,4,5,7	Modifications for 3500L floats	
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#### **GENERAL** SECTION 1.

This supplemental manual, applicable to those Cessna Model 185 Series airplanes equipped with Aerocet Model 3500 or 3500L Seaplane Floats, provides information and limitations not included in at the and/or

the basic FAA approved marki words "Not Applicable" (NA) related information may not be Flight Manual are not required referenced. The aircraft is to	appear in this supp e the same as that by the airplane ce	lemental manual, shown in the Ces rtification basis a	they are use ssna marking nd, therefore	ed to indicate that s and placards, , should not be
F	PERFORMANCE	- SPECIFICAT	IONS	
SPEED: NA				
CRUISE: NA				
RATE OF CLIMB AT SEA	LEVEL:	EXCEEDS	690 FPM	(CAR 3.85a)
SERVICE CEILING: NA TAKEOFF PERFORMANC	T. NIA			
LANDING PERFORMANC				
STALL SPEED (POWER C		CG).		
FLAPS UP:	60 KCAS	007.		
FLAPS DOWN:	55 KCAS			
MAXIMUM WEIGHT:				
RAMP (DOCK):	3	362 LBS.		
TAKE-OFF & LAND	DING: 3	350 LBS.		
EMPTY WEIGHT: SEE A				
MAXIMUM USEFUL LOA		AL WT. & BAL	ANCE FOR	RM FOR A/C
BAGGAGE ALLOWANCE				
IN AIRPLANE:		O CHANGE		
IN EACH FLOAT		00 LBS.	IENLOAD	INIC
(CAUTION: ASSU WING LOADING:	NA	15 PROPER WE	TEN LUAD	ing)
POWER LOADING:	NA NA			
RANGE: NA	NA.			
FUEL CAPACITY:	NO CHANGE			
OIL CAPACITY:	NO CHANGE			
ENGINE:	NO CHANGE			
PROPELLER:	NO CHANGE			
	A same			

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## SECTION 2. LIMITATIONS

#### CENTER OF GRAVITY LIMITS:

Center of Gravity Range: (inches aft of reference datum)

(+41.9) to (+46.5) at 3350 lbs. Max. G.W.

( $\pm$ 35.1) to ( $\pm$ 46.5) at 2200 lbs. or less with a straight line variation between points given.

# WEIGHT LIMITS:

Maximum Ramp (Dock) Weight:	3362 lbs.
Maximum Takeoff Weight:	3350 lbs.
Maximum Landing Weight:	3350 lbs.
Maximum Weight in Baggage Compartment:	NO CHANGE
Maximum Weight in Float Baggage Compartment	100 lbs. each.

#### AIRSPEED LIMITS:

	<u>KCAŞ</u>	<u>KIAS</u>	MPH CAS	MPH IAS
Never Exceed Speed (Vne)	179	182	206	209
Max. Structural Cruising (Vno)	143	146	164	168
Max. Maneuvering speed (Va)	115	117	132	135
Max, Speed with Flaps (Vfe) NO Cl	HANGE			

#### AIR SPEED INDICATOR MARKINGS:

The airspeed indicator shall be marked with a red radial line at 182 knts.

If the radial line is on the indicator glass, the glass and bezel should also have a white slippage mark.

Apart from the redline mark, the airspeed indicator markings are the same as shown in the basic markings/Flight manual. Due to differences in airspeed calibration and speeds with floats installed, the indicated stall speeds and maximum structural cruising speed vary slightly from airspeed indicator markings.

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#### PLACARDS:

. Aerocet P/N 35-70006 Placard is located on after part of console near water rudder handle in the "RETRACT" position:

> "WATER RUDDER ALWAYS UP **EXCEPT** WATER TAXIING"

Aerocet P/N 35-70005 Placard is installed on left side of cabin in full view of the pilot for either 3500 or 3500L float installation:

> "FLOATPLANE WITH AEROCET 3500 FLOATS

NEVER EXCEED SPEED:

182 KNTS, (209 MPH) (IAS)

MAX. MANEUVERING SPEED: 117 KNTS. (135 MPH) (IAS)

MAX. GROSS WEIGHT:

3350 LBS.

CG RANGE

(+41.9) TO (+46.5) AT 3350 LBS. MAX. GROSS WT.

(+35.1) TO ( +46.5) AT 2200 LBS. OR LESS WITH A STRAIGHT

LINE VARIATION BETWEEN POINTS GIVEN FOR WT. & BAL.

SEE LOADING SCHEDULE."

#### SECTION 3. EMERGENCY PROCEDURES:

Emergency procedures in the FAA approved airplane placards and /or Flight Manual generally apply except for airspeeds which may be different. Emergency landings on water should be done with water rudders up, aircraft slightly tail low on touchdown, and control wheel held full aft as the floatplane decelerates on the water. Emergency landings on land should be done with water rudders up, aircraft in a level attitude on touchdown, and the control wheel full aft after contact. If damage occurs to the floats causing compartments to flood, aggressively shift the weight (people & baggage) in the opposite direction of damage in order to balance the aircraft over the buoyant compartments.

SECTION 4. NORMAL PROCEDURES: (NOTE: THESE ITEMS SUPPLEMENT THE CESSNA NORMAL PROCEDURES- BE SURE AND FOLLOW THE CESSNA PROCEDURES EXCEPT AS NOTED BELOW)

Before Entering the Floatplane.

1. Inspect the floats and attachment for dents, cracks, punctures, etc.

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- 2. Remove rubber plugs (which serve as stoppers on the standpipe in each float compartment) and pump out any accumulation of water. Reinstall rubber stoppers with enough pressure for a snug fit. (If there is an excess of water, investigate the leakage)
  - 3. When checking engine oil level, use the side of the dipstick having two "X" marks. The lower mark indicates nine quarts and the upper mark indicates twelve quarts.

#### Before Starting Engine.

- 1. Water Rudder Operation "CHECK VISUALLY"
- 2. Water Rudders "DOWN FOR TAXIING"
- 3. Water Rudders "CHECK FREEDOM OF MOVEMENT & SECURITY

#### Takeoff

- 1. Water Rudders "UP" (retraction handle aft)
- 2. Wing Flaps "20 DEGREES" (second notch)
- 3. Control Wheel "HOLD FAR AFT INITIALLY"
- 4. Power "FULL THROTTLE & MAX RPM (advance slowly)
- 5. Control Wheel "MOVE FORWARD TO ATTAIN PLANING ATTITUDE"
- 6. Control Wheel "APPLY LIGHT BACK PRESSURE TO LIFT OFF"
- 7. Wing Flaps "UP AFTER OBSTACLES ARE CLEARED"

#### Before Landing

- 1. Water Rudders "UP"
- 2. Wing Flaps "DOWN"

#### Landing

- 1. Touchdown "SLIGHTLY TAIL LOW"
- 2. Control Wheel "HOLD FULL AFT THROUGH DECELERATION"

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<u>Balked Landing</u> - "RETRACT FLAPS TO 20 DEG. IMMEDIATELY AFTER APPLYING FULL POWER FOR GO-AROUND"

After Landing - "WATER RUDDERS DOWN"

<u>Securing Aircraft</u> - "FUEL SELECTOR TO RIGHT OR LEFT TANK POSITION TO PREVENT CROSSFEEDING"

#### **SECTION 5. PERFORMANCE:**

Airspeed Calibration - Essentially unchanged

**STALL SPEEDS:** 

POWER OFF, FORWARD CG, 3350 LBS

FLAPS UP:

60 KCAS

69 MPH CAS

FLAPS DOWN:

55 KCAS

63 MPH CAS

NOTE: ALTITUDE LOSS DURING STALL RECOVERY MAY BE AS MUCH AS 200 FEET.

#### **SECTION 6. WEIGHT AND BALANCE:**

The airplane equipped with Aerocet 3500 or 3500L Floats must be loaded in accordance with the limitations in Section 2. These are shown as an aircraft weight/moment envelope or an aircraft weight versus c.g. location chart on page 9.

Note: It is the responsibility of the airplane owner and pilot to insure that the airplane is loaded properly.

#### **SECTION 7. AIRPLANE AND SYSTEMS DESCRIPTIONS:**

In addition to the Aerocet 3500 or 3500L Float installation the aircraft must incorporate the Cessna approved seaplane kit. As a result of these installations, the floatplane is identical to the landplane with the following exceptions:

AEROCET MODIFICATIONS: Floats, incorporating a water rudder steering system, replace the landing gear. A water rudder retraction lever, connected to the water rudders by cables, is located on the cabin floor tunnel.

### **CESSNA MODIFICATIONS:**

- 1. An additional structural "V" brace is installed between the top of the front door posts and the cowl deck.
- 2. Additional fuselage structure is added to support the float installation.

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3. A centering spring assembly and cables are added to the rudder control system to improve stability in flight.

#### NOTE

If the floatplane is returned to the landplane configuration, these items need not be removed.

- 4. The floatplane has additional corrosion proofing and stainless steel cables.
- 5. Hoisting provisions are added to the top of the fuselage.
- 6. The left-hand cabin door is equipped with removable hinge pins for ease of door removal when loading large cargo.
- 7. Fueling steps and assist handles are mounted on the forward fuselage, and steps are mounted on the wing struts to aid in refueling the floatplane. Inboard fuel fillers are added when long range fuel tanks are installed.

#### **NOTE**

A reduction of approximately five gallons of usable fuel in each tank will result if inboard fillers are used to fill the long range fuel tanks.

- 8. A drain line is added to the front of the induction air manifold.
- 9. Floatplane placards are added.

**NOTE:** Refer to the appropriate Cessna Pilots Operating Handbook for other seaplane changes which apply to individual 185 models.

#### SECTION 8. AIRPLANE HANDLING, SERVICE, AND MAINTENANCE

Information not required

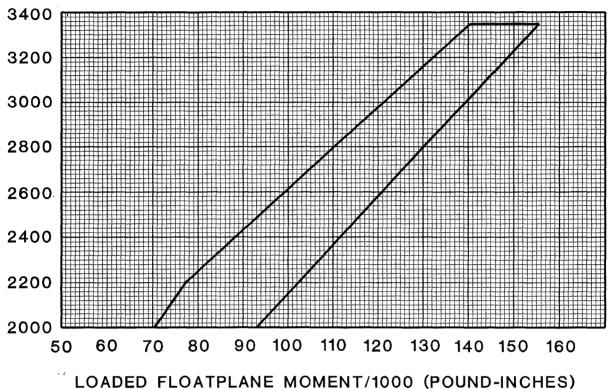
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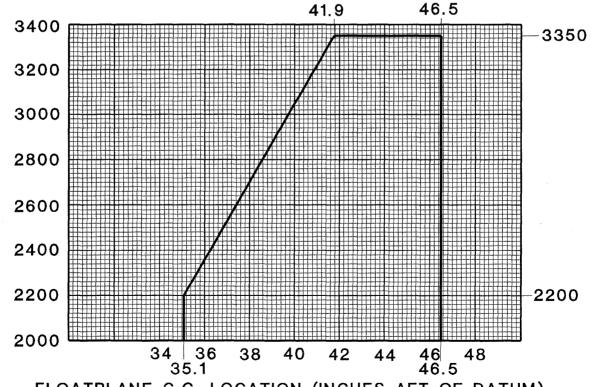


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41.9 to 46.5 at 3350 lbs 35.1 to 46.5 at 2200 lbs



11.0 16.5



FLOATPLANE C.G. LOCATION (INCHES AFT OF DATUM)

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