	Engineering Change Notice	ECN 1273
265 Shannon Lane Priest River, ID. 83856 (208) 448-0400		PAGE 2 of 2

Reason for Change: Clarification of Test Procedures, specifically differentiating between ON-GROUND and HOISTED methods.

Description of Change:

Section III Test Procedure

Added “There are two distinct methods of testing calibration. Use A below if available (preferred). Otherwise, use sequence in heading B.”

Kept all relevant information in §III. A for HOISTED aircraft test procedures.

Created §III. B. for ON-GROUND test procedures.

Various Pages (not revision-marked)

Added several Page and Section Breaks to help create better distinctions within the document. This increases the page count, thus making the revision effective on “ALL” pages.

-End-

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	1
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

REPORT A-31021

WOW, IN AIRCRAFT CALIBRATION

January 24, 2016

Written by Chuck Bovey

Revised February 28, 2018

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	2
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

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By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	3
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

REVISION RECORD

REV	DATE	DESCRIPTION	PAGES	BY
I/R	08/07/2015	Initial Release	All	
1	07/06/2017	<ul style="list-style-type: none"> • Edits to Section III.12 and III.15 (Test Procedure) details per ECN 1192 • Reformat to Aerocet document standard • Added Figure 1 and Plate 6 	All	LM
2	2/28/2018	<ul style="list-style-type: none"> • Per ECN 1273. [Clarification of Test Procedures; some formatting.] 	All	CB RB

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	4
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

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By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	5
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

TABLE OF CONTENTS

<i>Revision Record</i>	3
<i>Table of Contents</i>	5
<i>Table of Plates</i>	5
<i>Table of Figures</i>	5
I. Pre-test Preparation	7
II. Equipment Required	7
III. Test Procedure	9
a. If the Aircraft can be lifted off the ground and hung by the wings (hoisted)	9
b. If the Aircraft cannot be lifted off the ground and hung by the wings	10
IV. Calibration Harness with Switch	11
V. Strain gauge test procedure	17

TABLE OF PLATES

Plate 1 – Test Leads with Banana Plug and Shrink Tubing	11
Plate 2 – Adhesive Lined Heat Shrink Tubing over Terminals with Label on Switch Housing	12
Plate 3 – Test Leads with Terminal Switch	12
Plate 4 – Heat Shrink over Solder Joint	13
Plate 5 – Completed Calibration Harness	13
Plate 6 - Pin Diagram of WOW Box	15

TABLE OF FIGURES

Figure 1 - Pin Diagram of WOW Box	14
Figure 2 – Wire Harness Schematic	16

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	6
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

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By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	7
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

I. PRE-TEST PREPARATION

Before this procedure can be performed the 6650 floats and all the associated systems must be installed and operational on the Kodiak 100 aircraft, the center mid-floor panel must be removed to allow access to the WOW box assembly. The calibration can be done without lifting the plane by the wings to simulate in-flight conditions but accuracy and reliability are greatly improved by being able to simulate in-flight loads on the left mid strut.

II. EQUIPMENT REQUIRED

1. Digital Multi Meter (DMM) capable of reading 0.001 volts DC
2. Aerocet Calibration harness with switch
3. Small Jewelers screwdriver (.055 slotted)

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	8
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

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By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	9
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

III. TEST PROCEDURE

There are two distinct methods of testing calibration. Use A below if available (preferred). Otherwise, use sequence in heading B.

A. IF THE AIRCRAFT CAN BE LIFTED OFF THE GROUND AND HUNG BY THE WINGS (HOISTED)

1. Remove WOW box from mounting bracket in floor of aircraft
2. Remove 4 screws and cover from WOW box
3. Connect WOW box to aircraft harness
4. Connect Aerocet Calibration harness to the J1 test header on the WOW PCB (ensure that pin 1 of harness plug is in pin 1 of the header)
5. Connect Calibration Harness to DMM and set DMM to read "DC Volts" with a display resolution of at least 3 decimal places
6. Begin calibration with plane on the ground
7. Temporarily disable the stall warning by pulling the STALL circuit breaker
8. Turn on aircraft "Master Power Switch"
9. Turn on aircraft "Avionics Switch"
10. Set Calibration harness switch to position "A"
11. Adjust R24 on WOW PCB so that DMM is reading 2.700 VDC
12. Lift the plane and measure the position "A" voltage and record it (this voltage should be about 0.3 volts lower than the reading on the ground).
13. Calculate the mean or average voltage between the voltages readings in steps 11 and 12.
14. Set Calibration switch to position "B"
15. Adjust R6 so the meter reads the voltage calculated in step 13. (It should be near 2.550 volts)
16. Reset the stall warning breaker to enable the stall warning again.
17. With the plane off the ground, hanging from the wings, the audible stall warning should sound and the Flight HOBBS should be on and counting time.
18. With the weight of the plane on the floats, the audible stall warning should be silent and the Flight HOBBS should be off and not counting time.
19. Turn off Aircraft Master Power and Avionics Switches
20. Unplug Calibration harness from PCB, replace cover and screws on WOW box
21. Install WOW box into the aircraft floor bay mounting bracket
22. Recheck function of audible stall warning and Flight HOBBS

-End of Hoisted Procedure-

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	10
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

B. IF THE AIRCRAFT CANNOT BE LIFTED OFF THE GROUND AND HUNG BY THE WINGS

1. Remove WOW box from mounting bracket in floor of aircraft
2. Remove 4 screws and cover from WOW box
3. Connect WOW box to aircraft harness
4. Connect Aerocet Calibration harness to the J1 test header on the WOW PCB (ensure that pin 1 of harness plug is in pin 1 of the header)
5. Connect Calibration Harness to DMM and set DMM to read “DC Volts” with a display resolution of at least 3 decimal places
6. Begin calibration with plane on the ground
7. Temporarily disable the stall warning by pulling the STALL circuit breaker
8. Turn on aircraft “Master Power Switch”
9. Turn on aircraft “Avionics Switch”
10. Set Calibration harness switch to position “A”
11. Adjust R24 on WOW PCB so that DMM is reading 2.700 VDC.
12. Set Calibration switch to position “B”
13. Adjust R6 on WOW PCB so the meter reads 2.550.
14. Reset the stall warning breaker to enable the stall warning again.
15. With the plane off the ground, in flight, the audible stall warning should sound when appropriate and the Flight HOBBS should be on and counting time.
16. With the weight of the plane on the floats, the audible stall warning should be silent and the Flight HOBBS should be off and not counting time.
17. Turn off Aircraft Master Power and Avionics Switches
18. Unplug Calibration harness from PCB, replace cover and screws on WOW box
19. Install WOW box into the aircraft floor bay mounting bracket
20. Recheck function of audible stall warning and Flight HOBBS

-End of On-Ground Test Procedure-

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	11
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

IV. CALIBRATION HARNESS WITH SWITCH

The following pictures show an example of a harness constructed to assist in the calibration process. Prepare test leads with the male banana plugs on either end (Plate 1) (schematic BOM item 1 & 2) as follows:

Cut one red wire to 12” with banana plug on one end.

Cut two red wires to 12”

Cut one black wire to 24” with black banana plug on one end

Strip all cut ends .15”

Set aside the remaining parts of the test leads for the next harness



Plate 1 – Test Leads with Banana Plug and Shrink Tubing

Solder three 12” long red wires to switch. Center wire has banana plug on other end. Two outer wires have no connector. Place adhesive lined heat shrink tubing over terminals as shown.

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	12
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1



Plate 2 – Adhesive Lined Heat Shrink Tubing over Terminals with Label on Switch Housing

Place a label (Plate 2) on switch housing to identify switch position for test procedure. (A is on other side of switch) This example shows sharpie marking on white heat shrink. Note that when switch is placed in the A position, the center wire is connected to the near wire in this view. When switch is placed in the B position, the center wire is connected to the far wire in this view.

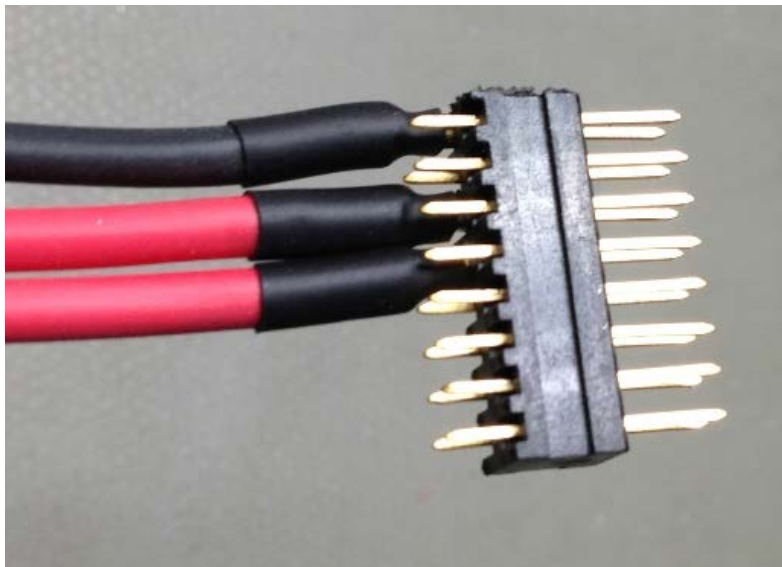


Plate 3 – Test Leads with Terminal Switch

Cut terminal strip (Item 4 in the schematic BOM) down to 16 positions. (Terminal strip shown in Plate 3) is different than that in the BOM but the function is the same). Lap solder two red wires to terminals per the schematic and place heat shrink over the solder joint (Plate 4). Wire A is connected to pin 10. Wire B is connected to pin 12. Lap solder black wire to terminal per the schematic and place heat shrink over the solder joint. Note that pin 1 is the lower near pin in this view. Pin 2 is the lower back pin in this view. The black wire is soldered to pin 16.

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	13
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1



Plate 4 – Heat Shrink over Solder Joint

After soldering wires to connector strip, use hot melt glue to fill in around the wires and unused pins to insulate the terminals and to build up structure for the adhesive lined black heat shrink tubing to form to. Place adhesive lined black heat shrink tubing over the body of the connector and wires to create a strain relief for the plug and wires. Label pin 1 with a silver or white sharpie.



Plate 5 – Completed Calibration Harness

Use tie-wraps or lacing to dress up and manage the wires as shown in Plate 5. The male banana plugs are on the end of black and red wires going out of the lower edge of the picture.

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	14
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

Figure 1 is the pin diagram on the circuit board of the WOW box.

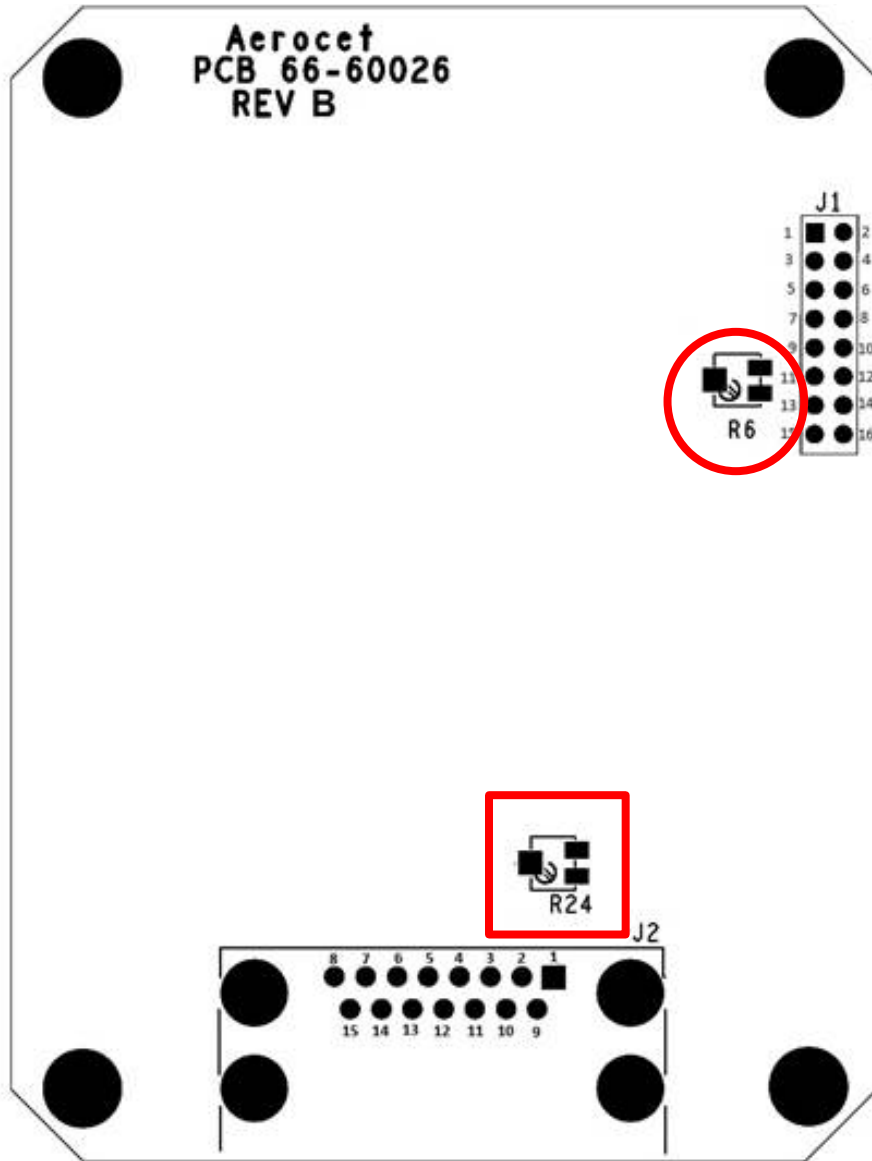


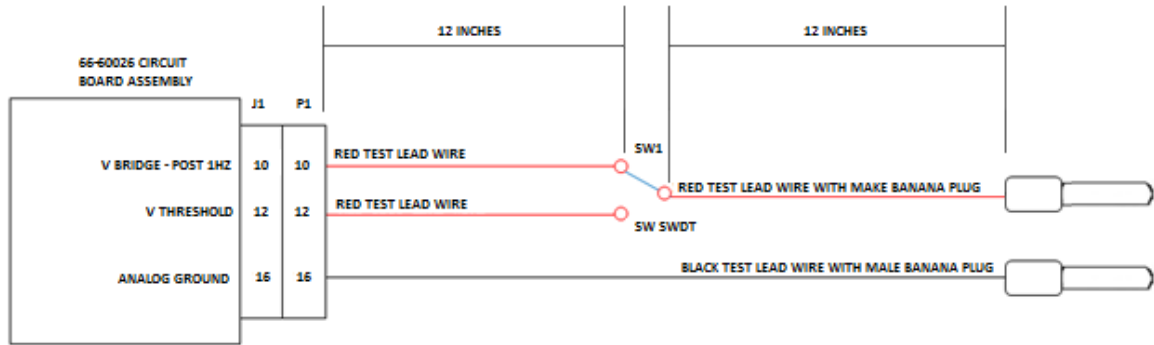
Figure 1 - Pin Diagram of WOW Box

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	15
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

Plate 6 is a picture of the open WOW box showing circuit board with pin configurations as outlined in Figure 1



Plate 6 - Pin Diagram of WOW Box



BILL OF MATERIALS

1. RED TEST LEAD 72 INCH: PAMONA ELECTRONICS PART NUMBER 8-72-2 (ONE PART MAKES TWO HARNESSES)
2. BLACK TEST LEAD 72 INCH: PAMONA ELECTRONICS PART NUMBER 8-72-0 (ONE PART MAKES TWO HARNESSES)
3. SWITCH, TOGGLE, SPDT: C&K COMPONENTS PART NUMBER 71015YZQE OR EQUIVALENT
4. PIN CONNECTOR, STACKING, 00 PCS: IE CONNECTIVITY PART NUMBER 4-146494-0 (ONE PART MAKES THREE HARNESSES)
5. HEAT SHRINK TUBING, 1" DIA X 3" LONG

Figure 2 – Wire Harness Schematic

By:	CB	Date:	1/24/2016	 PROCESS SPECIFICATION WOW, IN AIRCRAFT CALIBRATION	Page	17
CHKD:	CB	Date:	2/28/2018		Report No.:	A-31021
RVSD	RB	Date:	2/28/2018		Revision:	1

V. STRAIN GAUGE TEST PROCEDURE

The only way to verify the function and performance of the strut strain is to unload the strut either by lifting the aircraft by the wings so the weight of the floats is hanging from the fuselage as it is in flight or to fly the plane with the calibration harness and DMM connected to the WOW box. Prior to performing the following steps, verify that all the wiring and connections between the WOW box and the strain gauge are connected and not shorted to anything. Verify that there is no visible damage to the strain gauge installation on the strut. Verify that the WOW box is receiving aircraft power.

To verify that the strain gauge is working properly, perform the following steps:

1. While on the ground, perform or verify steps 1 through 11 of process specification A-31021.
2. If the aircraft can be lifted such that the weight of the floats are hanging from the fuselage, as they would be in flight, perform step 12 of A-31021. If the voltage drops by approximately 0.3 volts when the aircraft is lifted off the ground, the strain gauge is working properly.
3. If the aircraft cannot be lifted, leave the WOW box open, calibration harness and DMM still connected, and prepare the aircraft as necessary for a short test flight.
4. Monitor the DMM before, during and after takeoff. The voltage, with the calibration harness switch in position A, should drop by about 0.3 volts.
5. If the voltage does not change, the strain gauge will need to be replaced.

-END-